

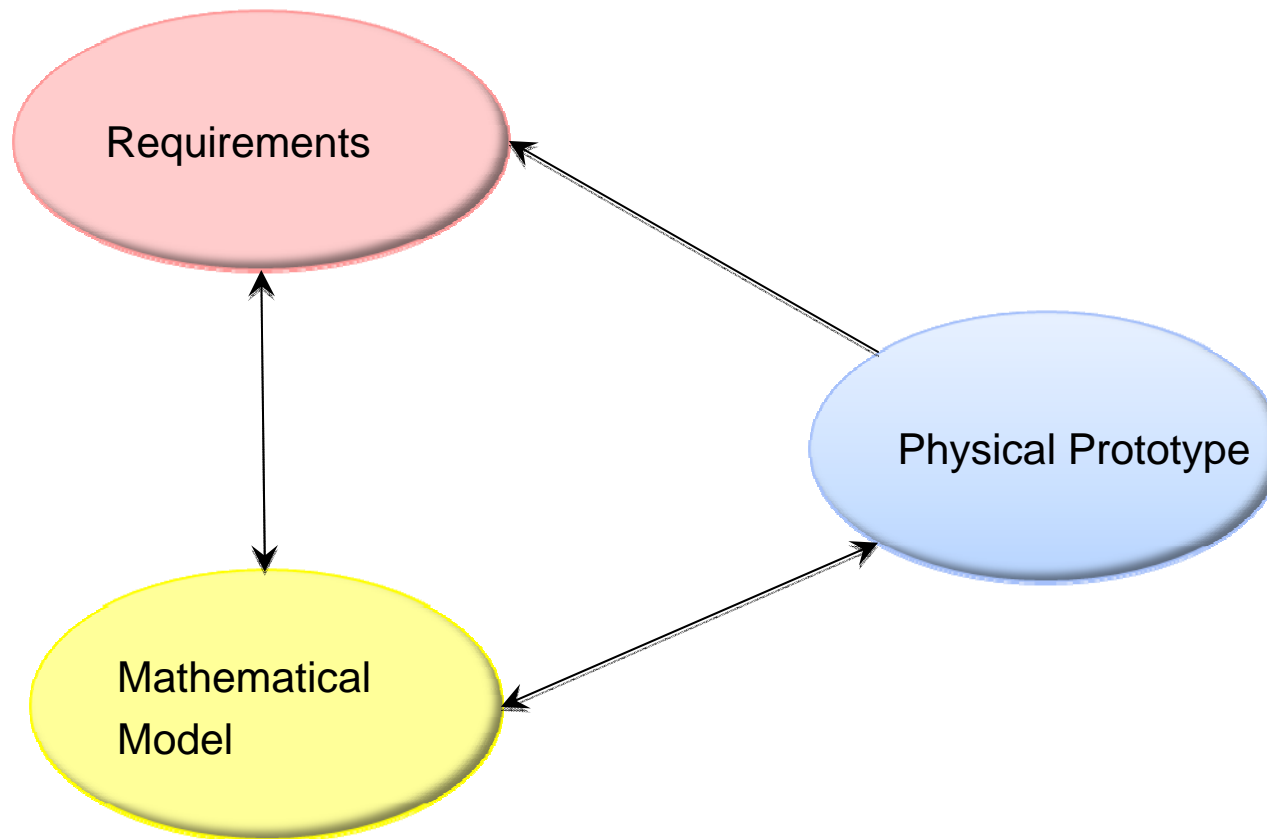
# Concept Realisation Laboratory

## *FluMeS and Machine Design*

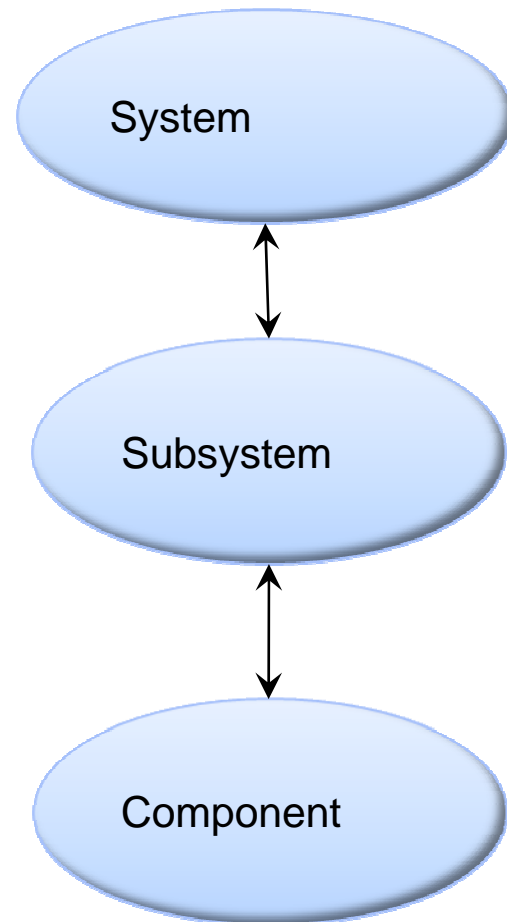


Petter Krus **IEI/FluMeS**

# Machine Design/Flumes Research philosophy



# Machine Design/Flumes Vertical Integration



# Increased Intelligence in Products

- New products tend to have more intelligence than before. This apply to a wide range of products. This is particularly true for the CRL application areas:
  - Cars
  - Aircraft
  - Construction Machinery
  - Industrial Manufacturing Systems



# Technologies

- These kind of systems are characterized by a close coupling between:
  - Mechanical system
  - Power transmission/Actuation systemSensors
  - Control System
- *This requires multi domain co-design.*

# Strong Trend Towards Multi Domain Analysis and Optimization

Design Variable	Actual Value	Target Value	Constraint	Structure Mass	Adaptive Mass	Propulsion Mass	Systems Mass	Max Mass	System Characteristic
10 Range	3733.21	4000.00	1.20	0.63	1.11	0.60	2.54	0.26	0.30
11 Lift-off distance	667.01	600.00	2.02	1.36	1.89	0.60	5.08	0.00	0.00
12 Landing distance	933.66	900.00	2.81	0.60	0.36	0.60	2.85	0.00	0.00
13 Fuel weight	10023.00	10000.00	0.96	0.36	0.60	0.85	0.00	0.00	0.00
14 payload	0.34	1.00	0.64	0.41	0.36	0.60	0.71	0.00	0.00
16 climb	291.42	300.00	0.96	0.26	0.11	0.60	1.22	0.00	0.00
18 Landing speed	30.76	70.00	1.00	0.26	0.11	0.60	1.42	0.00	0.00
17 Lift-off speed	46.24	70.00	1.00	0.26	0.11	0.60	1.42	0.00	0.00
19 Stall speed	76.27	80.00	1.00	0.26	0.11	0.60	1.42	0.00	0.00
19 Emissions	12000.00	10000.00	0.00	0.00	1.00	0.60	1.00	0.00	0.00
20 SFC	311.56	300.00	0.96	1.44	0.60	1.60	4.48	0.00	0.00
21 Cost	48426.15	40000.00	0.60	0.74	0.27	0.62	1.02	0.00	0.00

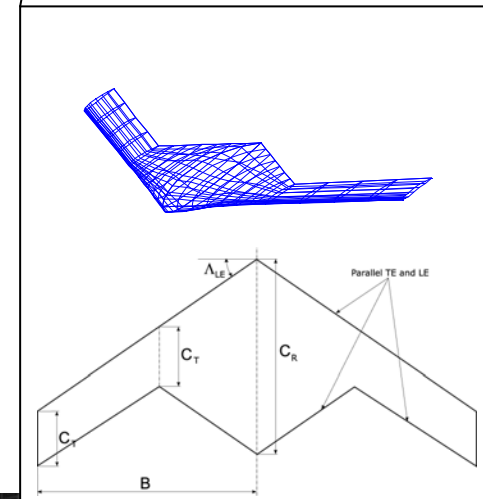
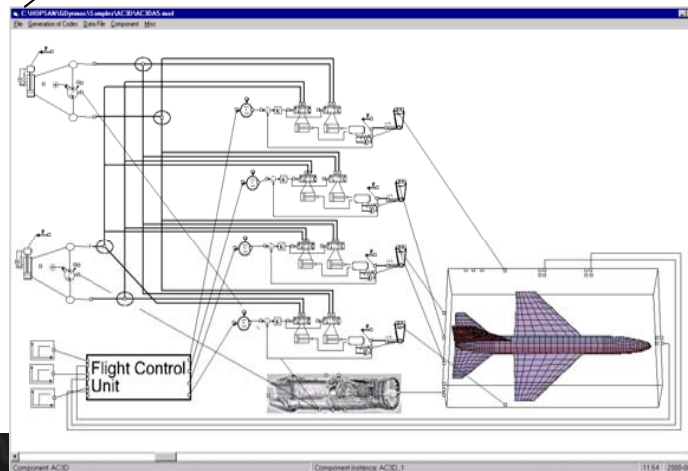
Spread sheet with design analysis and optimization tools

*Integrated system model*

*Simulation model*

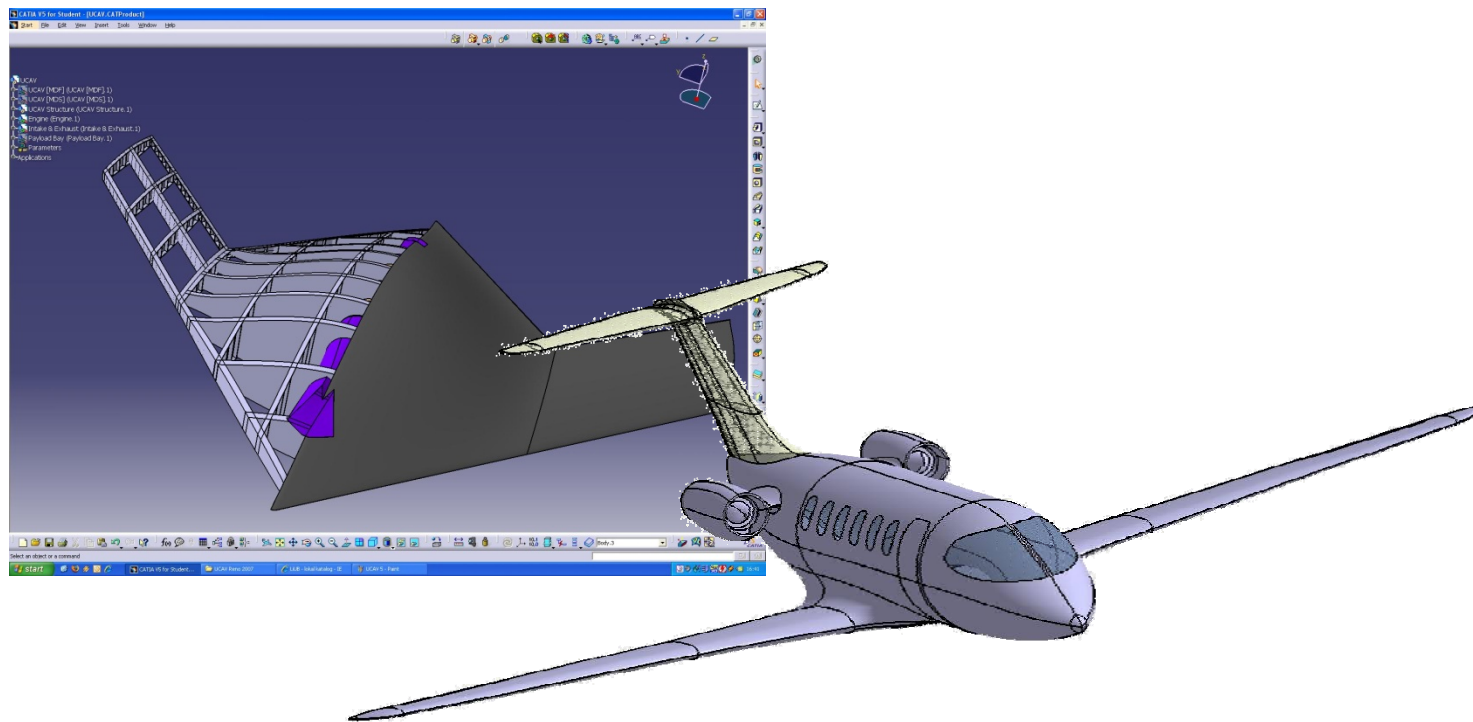
*Other analysis model*

Integrated system analysis of an aircraft with both an aerodynamic model and a simulation model

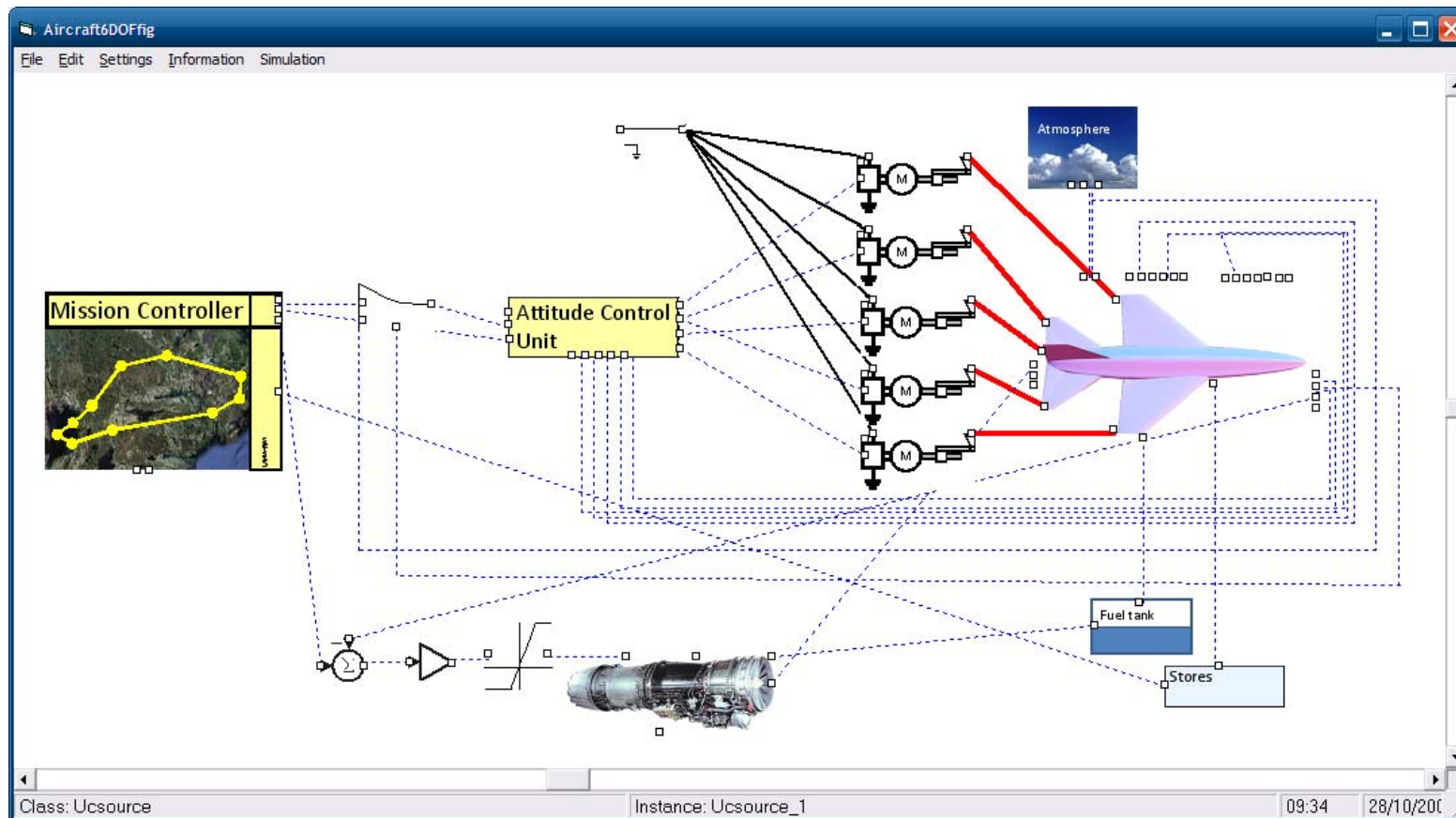


# CAD-Design Automation, Knowledge Based Engineering.

*Develop methodologies for next generation CAD-software*

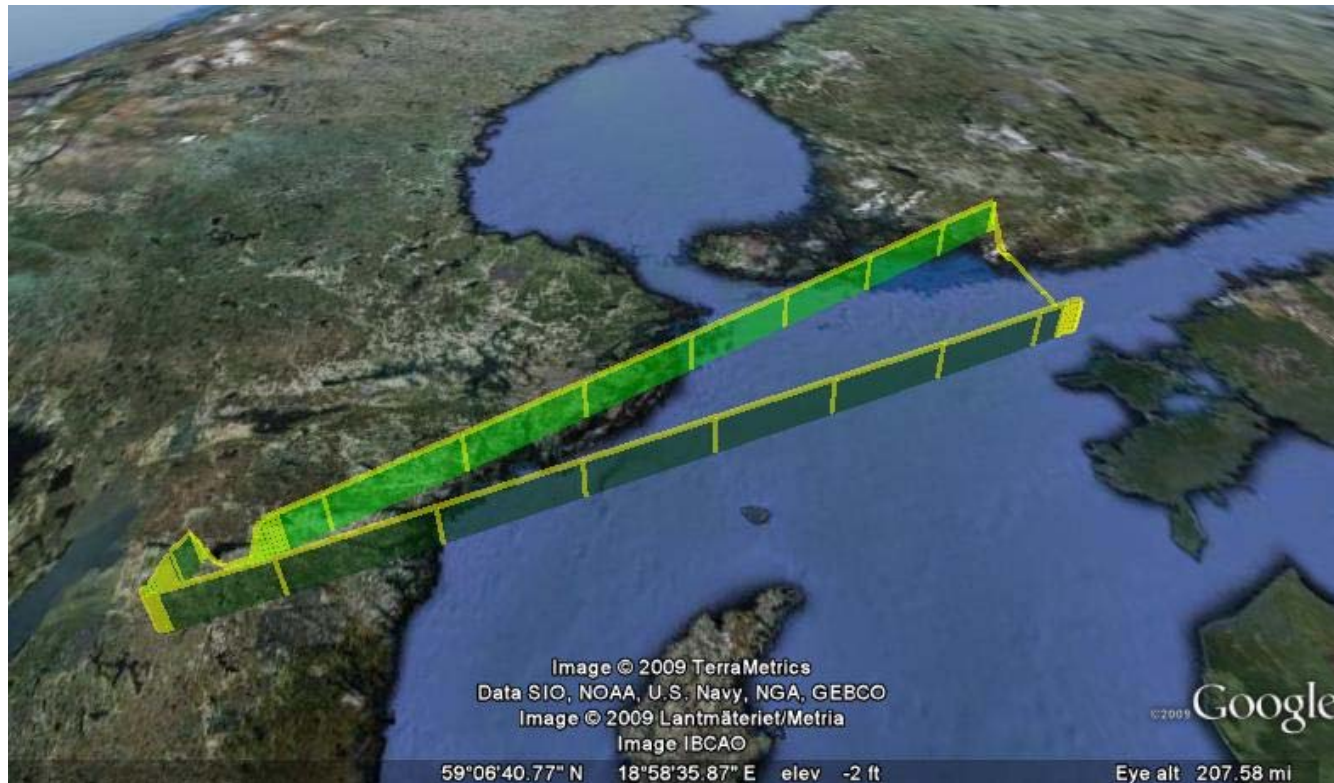


# Full Mission Simulation for Aircraft Design





## Towards Full Mission Simulation for Aircraft Design



**Figure 5. Simulated flight path (Furthest distance about 45 km from start point). Altitude scale is amplified 20 times for the plotting.**

(6000 sec simulated in 105 seconds (normal PC), time step 0.01 sec)

# Simulation in Heavy Vehicles

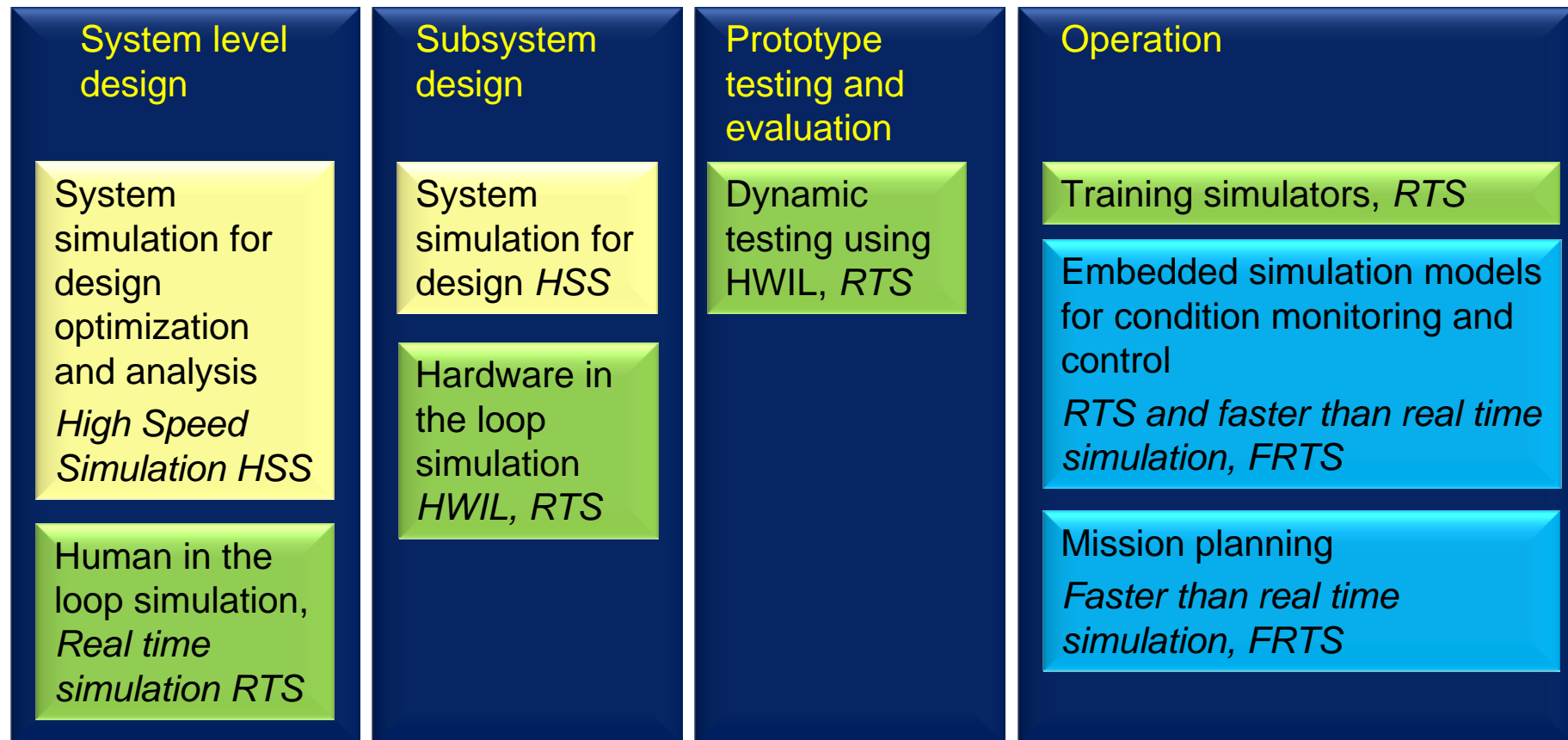


# Project Examples

- Energy Efficient Hydraulic System (Volvo CE, Parker)
- Hybrid Systems (Volvo CE)
- Sensorsystems for Trucks (Scania)
- Advanced Trajectory control (Scania)
- Aircraft Systems (Airbus Saab)
- Design Optimisation for Industrial Robots (ABB)
- High Speed Simulation (Volvo CE, Atlas Copco, CybAero, Prevas, National Instrument)

# High Speed Simulation for Product Design and Operation - HiPO

*-Using the same models throughout the lifecycle*



# Industrial Partners and Applications



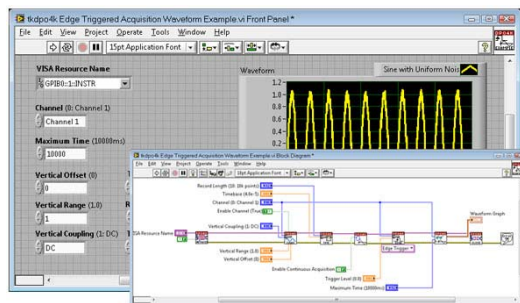
Helicopters  
*Cybaero AB*



Construction  
Machines  
*Volvo CE*

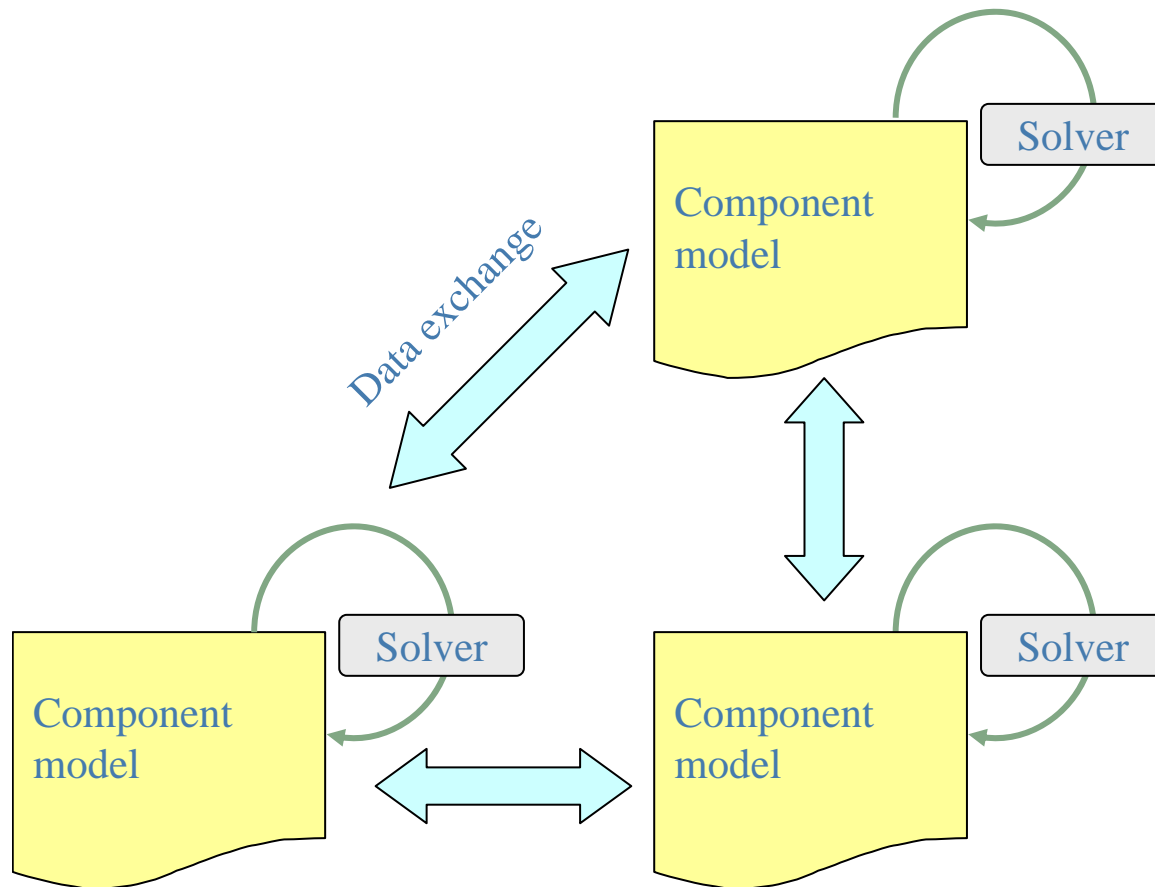


Rock drills  
*Atlas Copco*



Hardware in the loop  
systems  
*Prevas,  
National Instrument*

# Distributed modelling using transmission lines

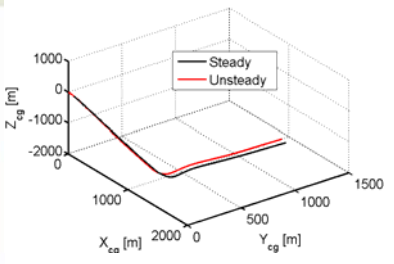
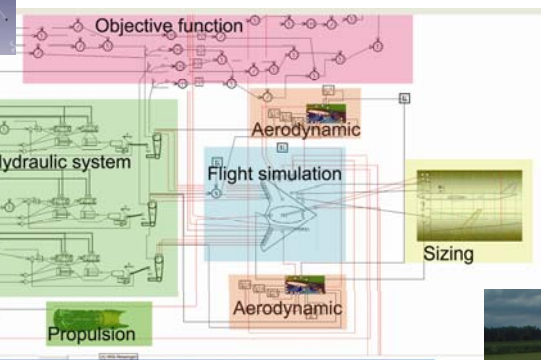
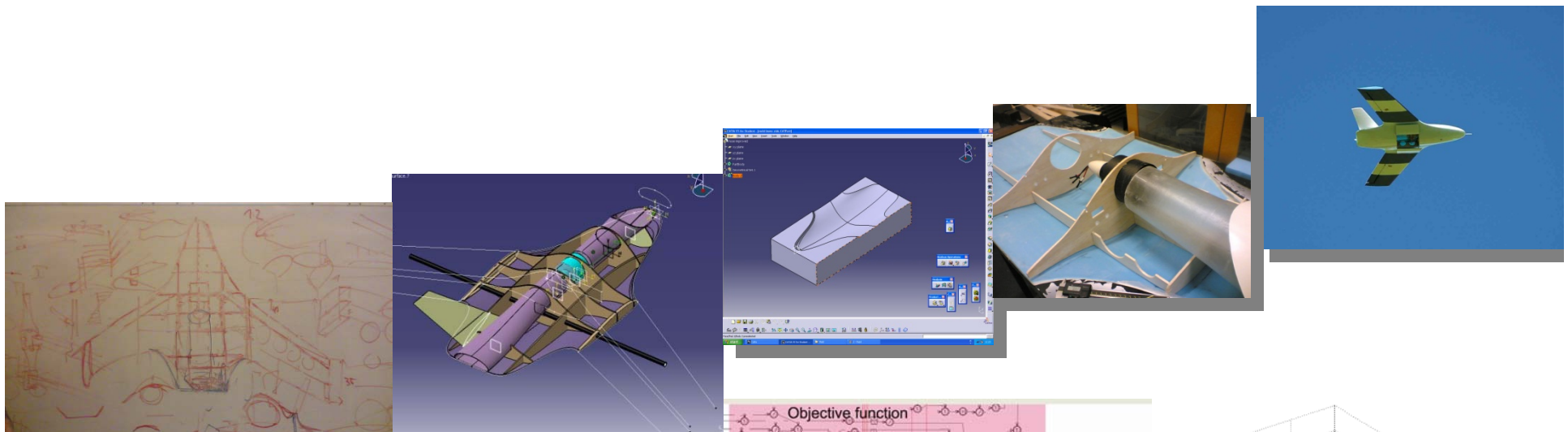


Maintain the physical structure of the system in the model

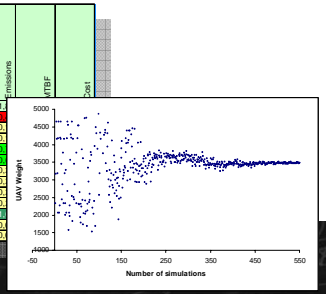
Use the finite signal propagation speed to numerical advantage

*Distributed* solver and allows for *distributed* processing

# From sketch to physical prototype in 5 months



System characteristics	Unit	Target value	Actual value
Range	km	5000.00	5490.87
Lift-off distance	m	500.00	393.93
Landing distance	m	500.00	104.19
Takeoff weight	N	60000.00	85865.23
Required weight quotient		1.00	0.98
Optimal cruise speed	m/s	100.00	146.68
Landing speed	m/s	70.00	25.48
Lift-off speed	m/s	70.00	30.33
Stall speed	m/s	80.00	50.55
Emissions		10000.00	18869.74
MTBF	hour	1000.00	7324.98
Cost	€EUR	40000.00	57060.64



# 2008 *GlobaLiTH*

## Light electric utility vehicle for development countries



13:34:31

				Motor power	Chassi cost	Safety weight	Battery weight		
System characteristics	Units	Target value	Actual value					System characteristics priorities	
Range	km	40.00	46.89	-0.07	0.14	-0.07	0.76		1.26
Acceleration (0-70)	s	8.00	5.44	0.06	-0.14	0.08	0.24		1.12
Top speed	km/h	100.00	135.76	-0.02	0.05	-0.02	-0.08		1.12
Recharge time	As Y	1.00	1.00	0.00	0.00	0.00	0.00		0.89
Handling	As X	1.00	1.00	-0.10	0.20	-0.11	-0.34		1.71
Safety level	As X	1.00	1.00	0.00	0.00	0.95	0.00		0.89
Running cost/km	EUR/km	2.00	0.11	0.07	-0.14	0.07	0.23		0.45
Emissions	As Y	1.00	0.00	0.00	0.00	0.00	0.00		1.12
Cost	SEK	50000.00	58804.53	0.44	0.47	0.00	0.09		0.45
				System parameters priorities					
				0.56	1.00	1.28	2.03		



# Winners of Formula ATA Electric and Hybrid Vehicles, Class 2, Rome 2009.

